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Mission Testing for Improved Reliability of CubeSats

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ABSTRACT

CubeSats are a small satellite technology that first emerged in the space sector in the early 2000s. Their low-cost, fast-delivery design has made space more accessible, leading to a significant increase in the number of satellites launched per year as well as the number of nations launching satellites. As an example, the Educational Irish Research Satellite, EIRSAT-1, is a CubeSat being developed by a student-led team at University College Dublin, aiming to launch Ireland's first satellite. Originally viewed as educational tools, the capabilities of CubeSats to satisfy a wide range of aims, from technology demonstration to Earth observation and scientific aims, is growing as these satellite platforms advance. However, following two decades of launches, an issue still facing their success as a disruptive technology is the high failure rate of CubeSats. This is a result of the CubeSat design philosophy where, in lieu of challenging resource demands, higher levels of risk are more acceptable compared to conventional space missions. In an effort to reduce these risks, this work presents an overview and highlights the benefits of a testing technique, known as Mission Testing. Mission Testing differs from other satellite testing in that flight representative conditions are simulated during the test. This aspect of the test mitigates risks not addressed elsewhere, such as risks related to the order and duration of operations, and also offers a number of unique benefits, such as operations training for the Mission Control Team. As part of this work, results of a survey disseminated to CubeSat teams are presented, where the aspects of Mission Testing considered most valuable and lessons learned in light of on-orbit operations are identified. How these results have helped shape the Mission Test plans for the EIRSAT-1 mission are also discussed and overview of the test plans are presented. In addition to the benefits for EIRSAT-1, this study aims to provide a resource for CubeSat teams in general, wanting to improve mission reliability through comprehensive Mission Testing.

Keywords: CubeSat, Testing, Reliability.

1. INTRODUCTION

The capabilities of nanosatellites have been growing rapidly since the first CubeSats were launched in 2000. Over one thousand launches later,^{1,2} CubeSats have proven themselves to be an extremely successful satellite platform. Their popularity is driven by their compact, standardised form factor, which is based on units of $10 \text{ cm} \times 10 \text{ cm} \times 10 \text{ cm}$ (i.e. 1U), weighing $\sim 1 \text{ kg/U}$.³ Larger CubeSats can be built measuring in multiples of 1U,

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typically up to 6U. Their small-scale, standardised design opens the possibility of using Commercial Off-The-Shelf (COTS) components and reducing typical barriers to entry, such as the high costs and lengthy development times associated with conventional space missions.

Until recently, CubeSats have been viewed primarily as tools for education, technology demonstration, COTS qualification, communications and Earth observation.⁴ However, their capabilities for certain scientific applications in astrophysics and planetary science are now becoming more widely considered,^{5,6} with major space agencies implementing CubeSats as elements within their programmes. For example, in 2018, as part of NASA's InSight Mars lander mission, two 6U CubeSats were used to provide a continuous communication link to the lander during its entry and descent to the Martian surface.⁷ As a result of these capabilities, as well as their accessibility and growing popularity, more nations are using CubeSats as a method of increasing their profile within the space sector.⁶

However, a factor still impeding the capabilities of CubeSats being more widely utilised is their failure rate; In 2018, $\sim 25\%$ of CubeSats launched failed to meet their primary objectives due to early loss of the mission.⁴ While this figure has significantly improved in the last two decades (the failure rate was $\sim 50\%$ in 2008^4), the development methodology around CubeSats, where less resources are invested to allow for increased accessibility, naturally leads to higher levels of risk and lower levels of reliability (i.e. likelihood of mission success). As CubeSat projects are often developed by university teams, the limited time, resources and experience available in this setting also likely impacts mission success rates.⁸ Overcoming this issue for the Educational Irish Research Satellite, known as EIRSAT-1, is a key driver behind this work.

EIRSAT-1 is a 2U CubeSat (Fig. 1) being developed by a student-led team at University College Dublin (UCD),⁹ and is set to be Ireland's first satellite. The design, build, test, and launch of the satellite are supported by the Education Office of the European Space Agency (ESA), under the 2nd round of the Fly Your Satellite! (FYS!) programme. As part of this opportunity, teams obtain access to state-of-the-art test facilities, receive guidance from ESA experts and, if certain milestones are achieved, are awarded a launch opportunity. EIRSAT-1 was one-of-six university-class missions selected in May 2017 to participate in this round of the FYS! programme, following the launch of 3 CubeSats in 2016 as part of round 1.¹⁰



Figure 1. Cutaway view of the EIRSAT-1 CubeSat. Orange markers label the parts of the satellite that are Commercial Off-The-Shelf (COTS) components. The remaining parts are custom-built components for EIRSAT-1.

As the first Irish satellite, the primary objectives of the EIRSAT-1 project are educational and focus on improving the capabilities of the national higher education sector in space science and engineering. In addition to educational aims, the mission will also perform the first in-flight test of three experiments. The ENBIO Module (EMOD) is a materials science payload hosting a thermal control experiment,^{11,12} Wave Based Control (WBC) is a software-based payload testing a novel attitude control algorithm¹³ and the Gamma-ray Module (GMOD) is a miniaturised γ -ray detector¹⁴ that will observe high-energy radiation from the most luminous electromagnetic explosions in the Universe, known as Gamma-ray Bursts¹⁵ (GRBs). The in-orbit demonstration of and science performed with these payloads aims to further highlight the capabilities of CubeSats, and potential of CubeSat constellations, for conducting scientific research.

To improve the likelihood of achieving these aims, the EIRSAT-1 team are developing robust testing techniques by assessing and building on the testing commonly used by other CubeSat teams to validate that the full satellite system, including the hardware and software of the space- and ground-segments, can perform the mission prior to launch. These testing techniques aim to mitigate the high levels of risk typically associated with CubeSats, as well as the risks associated in particular with aspects of the EIRSAT-1 project (i.e. student-led, no in-house experience from previous missions, multiple novel and complex payloads). This paper will present an overview of the testing performed by CubeSat teams, with a focus on Mission Testing. In Mission Tests, the in-flight operations, including nominal scenarios and contingency procedures for non-nominal scenarios, are simulated in a mission representative manner, and evaluated. As part of this discussion, key reasons for the different testing decisions taken by CubeSat teams will be considered. The apparent impact of these decisions on the on-orbit experience will also be reviewed. This work will draw on information gathered from literature, as well as from a survey disseminated to CubeSat teams with launch and on-orbit experience. The main details of a Mission Test plan that has been developed by the EIRSAT-1 team in light of this information will be presented. Future plans to assess the findings of this study following launch (in late $\sim 2021/22$) and operation of EIRSAT-1 will be identified. The aim of this work is to provide a resource for CubeSat teams, to assist the development of their Mission Testing and to improve the predicted reliability of CubeSats.

2. CUBESAT TESTING

Testing is a key and intuitive way to ensure that a system is capable of achieving the mission objectives. This can also be referred to as Verification & Validation (V&V), where testing is used as a means to *verify* that the mission requirements (a formal set of requirements set out at the onset of a mission that define the criteria for mission success) can be satisfied and to *validate* that the whole system can perform the intended mission, at the core of which is the requirements.¹⁶ The International Organisation for Standardisation (ISO) states that 'a certain set of tests is necessary to ensure the mission success of small spacecraft. Applying the same test requirements and methods as those applied to traditional large/medium satellites, however, will nullify the low-cost and fast-delivery advantages possessed by small spacecraft'. The standard ISO 19683:2017(E),¹⁷ Design qualification and acceptance tests of small spacecraft and units, was put in place specifically to address the reliability of small spacecraft through testing, to curb their failure rate while maintaining their appeal as alternatives to conventional space missions. This approach is adopted by many CubeSat teams (e.g. Endurosat¹⁸) and programmes (e.g. ESA's FYS!), where spacecraft testing and testing standards are followed but tailored for CubeSat purposes.

Similar to larger satellite projects, initial testing occurs during the development phase of a CubeSat. This can include unit-level software tests, where individual components of source code are tested, through to integration and hardware-in-the-loop tests. More formal testing and a significant part of the V&V process however then occurs at full system level, where all satellite parts are included in the configuration. This typically occurs first at a table-top 'FlatSat' level (demonstrated in Fig. 2) with some V&V of the system prior to 'stack' (Fig. 2) integration and testing.¹⁹ A range of tests can then be carried out on the full system as part of Functional Testing, where different, specific functions of the satellite and its subsystems (e.g. battery charging/discharging, RF transmission/reception, on-board data storage, etc.) are tested and then Mission Testing, where the performance of the full system is tested in a mission simulation^{*}. Additional testing performed by CubeSat teams includes

^{*}Note that some terms used in this work, such as 'Functional Testing' and 'Mission Testing' are commonly but not exclusively used by other CubeSat projects to describe similar testing.

vibration and thermal-vacuum (TVAC) testing, where parts of and/or the full satellite are subject to some of the physical extremes experienced during spaceflight.^{19,20}

All of the above-mentioned tests are important for the V&V process. In particular, Functional Testing is often used by teams as the main method for verifying the bulk of the requirements. As Functional Testing can be divided into parts, it is also commonly repeated, at least in part, throughout the development and testing of a CubeSat. However, only Mission Testing requires that flight representative conditions are simulated during the test. This aspect of Mission Testing has the potential to yield a surprising number of mission critical issues which other testing easily misses, and additionally offers a number of benefits beyond V&V of requirements, such as training of the Mission Control Team where some of the complexities of on-orbit operations are experienced through the simulation. Therefore, Sec. 3 of this work discusses Mission Testing for CubeSat projects in more depth to assess its impacts on mission reliability.



Figure 2. EIRSAT-1 hardware in a FlatSat (a) and stack (b) configuration.

3. MISSION TESTING

From the European Cooperation for Space Standardisation (ECSS), ECSS-E-ST-10-03C,²¹ Space engineering: testing, a Mission Test should:

- simulate the main in-flight situations and operations expected to occur over the entire mission profile (within the constraints of a ground-based test facility),
- include both nominal and contingency situations occurring in a realistic flight sequence,
- be uninterrupted (i.e. the satellite should remain powered on and operating throughout, without any interaction that would not be possible on-orbit), and
- use the final flight software as configured for launch.

Expanding on the V&V carried out via Functional Testing, Mission Testing primarily aims to *validate* the mission and to address potential issues related to the actual on-orbit operation of the satellite, such as issues related to the order and duration of operations. Testing with these aims is sometimes referred to as the 'test as you fly' approach. Another term with a similar definition that is commonly found in literature is 'Day in the Life' (DITL) testing. However, as the name suggests and in contrast to the ECSS definition where the 'entire mission profile' should be simulated, in some definitions of DITL (e.g. in Ref. 22), the primary aim is to demonstrate to launch authorities that the mission will meet requirements around deployables, timers, power inhibits, etc. during the initial stages (i.e. the initial ~24 hours) of the mission. A less frequently implemented approach is 'Week in the Life' (WITL) testing²³ which is closer to the ECSS definition of Mission Testing.

The high failure rate of CubeSats is a consequence of so-called Dead On Arrival (DOA) cases, where $\sim 20\%^{2,24}$ of CubeSats are never contacted following launch, and early mission loss. Early mission loss contributes to the high failure rate in particular over the first ~ 100 days of operations, during which the predicted reliability of a CubeSat (calculated from data on mission failures over time) drops by almost $\sim 20\%$.²⁴ The need for more testing is a common conclusion of studies assessing CubeSat reliability.^{8,23,24} While Functional Testing is of obvious importance to improving reliability, by ensuring that the satellite can at a minimum function as required for mission success, Mission Testing is also considered by some (e.g. Ref. 8) to be an essential step, as it allows issues related to the early operational phase of a satellite (i.e. during those first ~ 100 days) to be uncovered and addressed prior to launch. However, a search of the available literature on CubeSat testing does not reflect this perspective, with far fewer references to Mission Testing compared to other CubeSat testing techniques. Indeed, some adaptations of satellite testing standards for CubeSats, such as ESA's Tailored ECSS Engineering Standards for In-Orbit Demonstration CubeSat Projects,²⁵ actually exclude Mission Testing as a requirement, instead suggesting that some testing e.g. of 'operational mode transitions and safe mode recovery' scenarios should be covered through Functional Tests. Such an approach relates to the low-cost, fast-delivery advantages of CubeSats, as Ref. 8 states: 'it is one thing to identify [Mission Tests] as essential; it is another to carve out schedule and budget to carry out such tests'. These considerations lead to the main question to be addressed by this study viz. Could a lack of Mission Testing be a factor that significantly impacts the reliability of CubeSat missions?.

Reviewing the literature in which CubeSat Mission Testing^{26†} is discussed, one notable feature is the short time-frame defined for the test (i.e. a duration of 8 hours²⁶ is short when considering a Mission Test that should cover simulated situations occurring over the entire mission profile). However, Ref. 26 do state that their tests included both nominal and non-nominal situations that went beyond the first day of operations, which is in keeping with the ECSS Mission Test definition even though a highly accelerated mission life-cycle must have been simulated to suit the test duration. The information provided by Ref. 26 raises questions about the norms, if any, used by CubeSat teams when designing their Mission Tests, such as *what test durations are typically used?*, are tests repeated?, and how representative do the tests aim to be?. Building on the first question posed above in this section (i.e. could lack of Mission Testing be impacting the reliability of CubeSats), this study further considers that, if Mission Testing is performed by at least some CubeSat teams, what does this testing entail, where the aim is to address *what characteristics of Mission Testing are considered key by CubeSat Teams to improving mission reliability?*

While only limited literature on Mission Testing exists, it does not necessarily mean that such testing is not being carried out by many CubeSat teams. As this testing can be quite mission-specific, some teams may not see the need to publish details. Additionally, writing up publications requires additional effort and resources that may not be available within a team. Therefore, Sec. 3.1 of this work presents a survey, disseminated to CubeSat teams with launched missions, that was developed to better understand the extent to which Mission Testing is performed by CubeSat teams, and to assess whether the testing decisions made have an impact on mission reliability. By focusing in particular on the potential importance of Mission Testing, this work builds on other survey-style studies that collect information from those with on-orbit experience to investigate the leading reasons for CubeSat failures and improve their reliability. For example, Ref. 23 interviewed teams to build a set of recommendations for new projects, Ref. 24 have combined publicly available data with survey results to assess known and believed reasons for on-orbit failures as well as the time-dependence of these failures, and Ref. 27 have used surveys to identify design decisions that impact the likelihood of mission success.

[†]While the term 'Mission Test' may not specifically be used, similar testing aims and methods are described.

3.1 Mission Testing Survey

The survey, titled *Mission Testing vs. On-Orbit Mission Operations*, was developed to test the hypothesis that extensive Mission Testing prior to launch is an important step towards improving the reliability and performance of a CubeSat mission. Participants were required to have good general knowledge of the pre-launch testing and (if launched successfully) the post-launch operations of the satellite. The survey was composed of three sections:

- **Background** to establish the setting in which the CubeSat project was developed (i.e. university, industry, research organization, etc.), as this may have an impact on the approach to testing and the methods used,
- **CubeSat Testing** to determine if Mission Testing was carried out and, if so, to assess the extent (duration, repetition, conditions simulated, etc.) of the testing performed, and
- **Mission Operations** to determine if the mission's objectives were achieved on-orbit and if so, whether this can be attributed to the risks mitigated via Mission Testing.

The final page of this survey also asked participants to consider their main lesson learned that they would apply to future test campaigns.

In addition to questions with defined single and multiple choice answers, participants were also given the opportunity to elaborate on answers via *More Information* boxes following each question. The full survey, including all questions and the question logic applied based on the participants responses, is presented in Appendix A. This survey was conducted using the website JotForm and can be accessed at https://eu.jotform.com/form/201262644487053. Over the course of ~6 months, 8 participants anonymously took part in this study.

3.1.1 Results

The main findings of the survey are as follows:

• Background:

 The primary setting in which the CubeSat projects associated with the survey responses were developed are shown in Fig. 3. This figure also shows where participants indicated that their project involved cross-sector collaboration.



Figure 3. Primary setting in which the CubeSat projects were developed. Stacked bars represent cases where projects were supported by additional involvement from industry, university and/or a space agency.

• 6 out of 8 participants indicated that their projects had primary objectives that were focused on multiple goals (see Tab. 1), where technology demonstration was the most observed objective, followed by education.

Participant ID	Comms.	Tech. Demo.	Earth Obs.	Education	Science	Other
1		1		1		
2	1	1	1		1	
3			1			
4				1		
5		1		1	1	
6		1	1	1		
7		1	1	1	1	1
8		1			1	

Table 1. Mission Objectives.

• CubeSat Testing:

- 6 out of 8 participants indicated that, given the ECSS definition of a Mission Test, Mission Testing was performed by their CubeSat team.
- For the 2 participants who indicated that Mission Testing was not performed, other system testing, such as Functional and End-to-End Tests, providing 'satisfactory tests of the mission' and 'time constraints' were listed as the main reasons for not carrying out the test.

Note that while they indicate that no Mission Test was performed, both participants still emphasised the importance of testing in their responses, where one of the participants goes on to clarify that 'tests very similar to the Mission Test were performed that covered all of the criteria separately - the full test would not have provided a significant advantage.'

- From the 6 surveys where participants indicated that a Mission Test was performed:
 - * The number of Mission Tests performed by the CubeSat teams ranged from 1 to more than 10. A similarly wide range of responses was also received for the duration of each test performed, where participants indicated test durations of a couple of hours to 1-2 weeks (see Tab. 2).

Table 2.	Number c	of Mission	Tests	carried	out	by t	the	CubeSat	teams	and	$_{\rm the}$	test	durations	used.
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Participant ID	Number of tests performed	Approximate duration of each test
1	1	Couple of hours
2	>10	Couple of hours
3	5	12 hours
4	3	2-7 days
5	1	1-2 weeks
6	2	1-2 weeks

* These durations were primarily chosen as 'longer test durations would have been impractical'. Most (4 out of 6) participants also indicated that the test durations used were seen as 'sufficient to meet the aims of the test'.

- * All Mission Tests performed included at least the main, if not most, nominal situations that were foreseen for the mission. Non-nominal situations were also simulated by all but one of the teams, where the survey participant from this team indicated that simulating non-nominal situations as part of their tests was not possible given their 'limited amount of testing capability'.
- * Additionally, participants indicated that the Mission Tests performed by their CubeSat teams incorporated some realistic aspects of on-orbit operations, such as:

Participant ID	in the second se	C. C	the second secon	Jose Contraction of the second	A Contraction of the contraction
1		1	1	1	1
2	1	1	1	1	1
3	\checkmark		1	1	1
4		1	1	1	1
5		1	1	1	1
6					1
Simulated by:	2 teams	4 teams	5 teams	5 teams	6 teams

Table 3. Aspects of real on-orbit operations simulated as part of Mission Testing.

• Mission Operations:

- From the 6 surveys where participants indicated that a Mission Test was performed:
 - * 4 of the 6 CubeSats achieved the primary objectives of their mission, with operational lifetimes (at the time of the responses) ranging from 140 days to 4 years.
 - * Participants from all 4 teams indicated that their on-orbit experience was 'mostly nominal' and only 'rarely' or 'occasionally' encountered situations not considered for testing, primarily during early phase operations. Two participants went on to indicate that they believe their tests could have been improved to cover these situations, while the remaining 2 participants were unsure.
 - * The participants also indicated that in comparison to their on-orbit experiences their Mission Tests were 'somewhat' (3 participants) and 'very' (1 participant) representative.
 - * All four participants additionally considered that, in light of other tests that occurred throughout the CubeSat project, Mission Testing in particular played a 'significant' role in the success of the mission.
 - * The remaining 2 CubeSats for which Mission Testing was performed did not achieve the primary objectives of their missions due to launch failure and suspected DOA (or soon after). The participant from the latter mission indicated that improved Mission Testing could have reduced the risks that potentially led to the CubeSat not achieving its primary mission objectives.
- From the 2 surveys where participants indicated that no Mission Testing was performed:
 - * The primary mission objectives were achieved for 1 of the CubeSats, which had an operational lifetime of ~ 2 years. The survey response associated with the other CubeSat project indicated that the mission was not yet launched at the time of the response.

Key results from this section of the survey are summarised in Fig. 4.



Figure 4. Chart comparing the numbers of participants who stated their CubeSats did and did not achieve the primary mission objectives with whether Mission Testing was performed for the CubeSat mission.

- Lesson learned:
 - A number of the participants emphasised the importance of simulating on-orbit operations as realistically as possible during testing (e.g. 'be as operationally representative as you can' and 'drive Mission Tests from high-level operations requirements and let the operations team lead').
 - Including failure scenarios as part of testing to test redundancy and recover systems was also advised.

3.1.2 Analysis & Discussion

An initial observation from the results presented above is that, in contrast to what the lack of literature on Mission Testing might suggest, the survey responses indicate that Mission Testing, in at least some capacity, may be performed by many CubeSat teams (75% of the participants indicated that Mission Testing was performed for their mission). The responses further show that Mission Testing is a testing technique that is considered valuable by CubeSat teams regardless of the project setting (i.e. not just a testing technique favored for industry-led projects which typically have more resources), as 50% of the CubeSats for which Mission Testing was carried out were student-led. The remaining projects were industry-based CubeSat projects.

While 75% of the survey participants indicated that the ECSS definition of Mission Testing was conducted by their teams and although the main characteristics of the test are defined in ECSS and other (e.g. ISO) standards, another clear finding from the responses is that Mission Test execution varies significantly between different CubeSat teams. This can be noted throughout the results presented above but is particularly obvious for the durations of the Mission Tests carried out as well as for the number of times each team repeated Mission Testing (see Tab. 2). Tab. 4 further emphasises this point by showing the total time spent Mission Testing noted in this survey may somewhat be explained by that fact that no participant indicated that 'a recommended test duration was used', which was one of a number of multiple-choice answers to the question 'Why was this duration chosen for the Mission Test (s)?'. This suggests that each team independently defined their own the Mission Test characteristics, such as the test duration, without strong instruction from standards, requirements, etc. This all suggests that little-to-no norms around Mission Testing are shared and used by CubeSat teams when designing their tests. Is this due to the lack of literature, lack of CubeSat-specific testing standards or, as the responses of this survey allude to, is it primarily driven by the limited amount of time and resources typically invested into a CubeSat project?

Participant	Total testing time
ID	
1	Couple of hours
2	20+ hours
3	2.5 days
4	6-21 days
5	1-2 weeks
6	2-4 weeks

Table 4. Total time each team spent Mission Testing (developed from Tab. 2).

Although, by definition, less time and resources are invested into CubeSat projects compared to conventional spacecraft development projects, this survey highlights that some areas of satellite development, such as comprehensive system-level testing, including Mission Tests, are needed to improve the likelihood of mission success. The negative impacts of not investing time and resources into such tests are most clearly observed for the the CubeSat team where only \sim hours in total were spent Mission Testing (i.e. the first team listed in Tab. 4). The participant associated with this CubeSat mission indicated that schedule constraints (i.e. 'a launch deadline') led to this test duration being used. While one of the main issues with a short Mission Test duration is that risks associated with age-related failures are not well mitigated, this participant identified another issue in that the team were highly limited in what could and should be tested within the time frame (e.g. the participant indicated that it was not possible to simulate any non-nominal situations in this time). Communication with this CubeSat was never confirmed following launch. The participant goes on to suggest that improved Mission Testing could potentially have prevented the loss of their mission. This survey response clearly demonstrates how the quality and characteristics of the Mission Testing performed by teams can impact the likelihood of mission success and more specifically shows that a Mission Test duration of a couple of hours is not sufficient to improve mission reliability. This response also highlights that, testing decisions, such as the amount of time allocated to a test, must be driven primarily by the test objectives and the realistic timeline in which these objectives can be achieved rather than schedule demands, for the test to have a positive impact on the reliability of the mission.

As the second team listed in Tab. 4 experienced launch failure, the remaining 4 entries show the testing times for the teams who performed Mission Testing and achieved their primary mission aims. The participants from all 4 teams indicated that they had 'mostly nominal' on-orbit experiences and only 'rarely' or 'occasionally' encountered issues not considered for testing. Therefore, no further comparisons between the type of testing carried out by these teams and their on-orbit successes/failures can be drawn from the survey responses. However, it is worth noting that all of these teams performed, in total, at least ~days worth of Mission Testing. This test duration seems more consistent with what would be the expected duration required to achieve the objectives of the ECSS Mission Test, where both nominal and non-nominal situations expected to occur over the entire mission profile are simulated.

Participants' responses to the lessons learned question emphasised the importance of simulating on-orbit operations as realistically as possible during Mission Tests. This is consistent with the main goal that Mission Testing aims to achieve for a mission beyond other testing where, as stated above, one of the primary aims is to address potential issues related to the actual on-orbit operation of the satellite. This feedback is also largely supported by further results from the survey, presented in Tab. 3, showing that key aspects of real on-orbit operations were simulated by many of the CubeSat teams who performed Mission Testing. One noteworthy and conflicting result from Tab. 3, however, is that only 2 out of the 6 teams who performed Mission Tests incorporated limited 2-way communications (i.e. short communication 'windows' separated by long durations that result from the CubeSat's orbit) between the space and ground segments into their testing. The most

obvious and likely reasons for not simulating limited communications as part of a Mission Test are related to time, as simulating realistic communication constraints (only \sim tens of minutes of interaction with the spacecraft per day) will increase the required test duration significantly. However, when the aim is to perform a test that is as representative as possible, a strong argument can be made that this aspect of real on-orbit operations (or, at least, a reduced version of this aspect) is as essential as any of the other aspects listed when considering the risks that are mitigated and the uncertainties that are addressed - e.g. is the spacecraft capable of performing the mission with only ~minutes of contact with the ground station per day? How robust is the spacecraft to surviving anomalies without immediate intervention from the Mission Control Team? Can enough of the data generated onboard be downlinked during the short communication windows to meet the mission requirements? Furthermore, the decision not to simulate realistic communication constraints can actually impact how representative other aspects of the Mission Test can be. For example, as the spacecraft will be transmitting data more frequently without long breaks between communication windows, the spacecraft's power consumption will not mimic the on-orbit behaviour of the system. Additionally, if there is no constraint on communication, the Mission Control Team and protocols will not experience the many stresses related to fixed and limited communication times. This may explain why 3 out of the 4 participants from teams who performed Mission Testing and later operated their spacecraft on-orbit indicated that Mission Testing was only 'somewhat' representative of real operations, and may also explain the main reason behind the lessons learned from each.

As only 2 participants indicated that Mission Testing was not performed for their mission, and as only one of these participants indicated that their CubeSat had been launched at the time of the response, the impacts, negative or otherwise, of *not* performing Mission Testing on mission reliability cannot be shown from this work. This result would require more survey respondents (see Sec. 5). However, a clear finding from the current set of responses is that CubeSat teams who perform Mission Testing rate it highly as a tool to improve the likelihood mission success (all participants for which the primary aims of the mission were achieved marked Mission Testing as a 'significant' contributor to their success and, for the case where the aims were not achieved, the participant believed that improved Mission Testing could have prevented the loss of their mission). Therefore, the findings of this survey support the earlier hypothesis that Mission Testing is an important step to improving the predicted reliability and performance of a mission prior to launch.

3.2 EIRSAT-1 Mission Testing

EIRSAT-1 is set to be Ireland's first satellite. As such, recommendations in the form of standards, requirements and lessons learned are extremely valuable to the EIRSAT-1 team as in-house experience and expertise in satellite projects are only now being fostered. For this reason, first-time projects, like EIRSAT-1, also benefit in particular from being involved in a structured programme such as ESA's FYS!, where teams must follow the programme's requirements, which have been set out by CubeSat experts, and further recommendations are readily available. Some of the main recommendations related to Mission Testing that the EIRSAT-1 team have received as part of FYS! are given in Sec. 3.2.1.

Support from those with expertise is especially important for testing and in particular for Mission Testing, as the team try to mitigate failures and simulate scenarios that have not yet been experienced first-hand. Therefore, in addition to the support received as part of the FYS! programme, the EIRSAT-1 team have also aimed to build knowledge from other CubeSat teams with different development, test, launch and operations experiences. This is a leading driver behind the survey presented in Sec. 3.1. The impact of the survey results to date on team's test plans are mentioned throughout the next section, where the main characteristics of the EIRSAT-1 Mission Test campaign and the rationale behind these characteristics are discussed.

3.2.1 Test Plan

Number of Mission Tests	2
Duration of each test	~ 3 weeks
Situations simulated	 Most nominal situations expected to occur during the mission lifetime[‡] Main-to-most non-nominal situations expected to occur during the mission lifetime[‡]
Additional aspects simulated for the test	 Representative physical state of the spacecraft Limited communication windows separated by long durations Realistic power constraints Operations via the Mission Control Team and protocols Interaction with the space segment solely via radio[§] Use of the ground segment hardware and software[§]

Table 5. Main characteristics of the EIRSAT-1 Mission Test campaign.

Mission Testing is a requirement for CubeSat teams participating in the FYS! programme, as FYS! consider it to be an essential test to perform for mission validation prior to launch. The programme recommends that Mission Tests should last a minimum duration of 1-2 weeks, uninterrupted. During this time, it is expected that both nominal and non-nominal situations are simulated, starting from launch through to the 'normal' mission operations phase. Similar to the feedback from the survey, FYS! stress the importance of simulating realistic onorbit conditions as best as possible during Mission Testing - 'test as you fly!'. Use of the mission's real operational procedures, communication over a radio link, use of the actual ground segment set-up, etc. are all encouraged. In light of these recommendations, a Mission Test duration of \sim 3 weeks was chosen by the EIRSAT-1 team, as this was considered sufficient to perform the test described in this section.

In an effort to be representative as possible, the EIRSAT-1 Mission Tests will include some realistic aspects of on-orbit operations, which are listed in Tab. 5. The aspects in this table include those provided as multiple-choice options to the survey participants but additionally include that the physical state of the spacecraft at different points in the mission is simulated as well as the simulation of realistic power constraints. For the former, "launch" and "deployment" into orbit will be conducted in a representative manner with respect to the spacecraft's power systems, where deployment switches will prevent the CubeSat from powering on in its launch configuration until deployment. Furthermore, tumbling of the spacecraft at ~tens of degrees/second following deployment, and for later attitude control tests, will be simulated using a rig developed to rotate the CubeSat.²⁸ For the latter, software to control the power supply responsible for charging the spacecraft's solar cells for ~45 minutes every ~90 minutes (based on a spacecraft in Low Earth Orbit). As limited 2-way communication[¶] is also being simulated as part of the test, the rate of battery discharge will similarly be representative, as the spacecraft's radio will only be transmitting large quantities of data during ~minutes-long communication windows which are separated by ~hours.

A variety of nominal and non-nominal situations will be simulated during the ~ 3 weeks of Mission Testing, starting from pre-launch preparations where the CubeSat is prepared for shipment to the launch provider. Examples of the situations to be considered are given in Fig. 5. Nominally, multiple of these situations will be simulated during each testing day. However, non-nominal situations and any anomalies encountered (i.e. unexpected events which are being treated as part of the Mission Test - advice from the survey: 'Don't assume

[‡]Given what is possible to simulate within the constraints of a ground-based test facility

[§]Simulated in-full for only 1 of the Mission Tests due to level of readiness.

 $[\]$ Although the survey showed that communication constraints were not commonly simulated by CubeSat teams for Mission Testing, the EIRSAT-1 team have decided this is an important aspect to simulate for the reasons listed here and in Sec. 3.1.2



Figure 5. Situations simulated as part of EIRSAT-1's Mission Test, where SC = spacecraft, GS = Ground Station, ITU = International Telecommunication Union and FDIR = Fault Detection, Isolation and Recovery.

that testing covers all scenarios, test what happens when things break down unexpectedly and see if the satellite can recover.') may require more time to resolve.

Following the main Mission Test, the EIRSAT-1 team have additionally decided to test a series of worstcase, non-nominal Launch and Early Operations Phase (LEOP) situations^{||}, where, for instance, the CubeSat is launched with low battery or where an anomaly occurs that causes the spacecraft to reboot several times prior to antenna deployment. These situations are being considered in the context of a full Mission Test simulation, rather than e.g. as part of a unit test of the post-launch software or a Functional Test of the antenna deployment mechanism, to further mitigate risk to mission success during a highly critical point in the mission. The reasoning for this additional testing is also supported by the fact that many of the participants from the survey indicated that it was during LEOP, or soon after, that situations not considered for testing where encountered.

The EIRSAT-1 team will perform Mission Tests twice, on 2 separate fully integrated models of the spacecraft known as the Engineering and Qualification Model (EQM) and the Flight Model (FM) (see Sec.4 for more information of this model philosophy). One of the main advantages to repeating the test relates again to this being a first-time project, as the EQM test will allow the EIRSAT-1 team to practice the Mission Test and develop it further for the FM. Repeating the test is also essential for the EIRSAT-1 team as some aspects of the Mission Test will only be possible for the FM test campaign (e.g. see Tab. 5). While preferably these aspects would be included in both of the Mission Tests to be performed, the associated risks and resources were assessed by the EIRSAT-1 team and the decision that the test is performed at least once *in full* was seen as satisfactory to meet the aims of Mission Testing, particularly given the resources being invested into each test campaign.

^{||}These are not possible to simulate within the one, main Mission Test which must be 'uninterrupted' as these nonnominal situations require that the spacecraft is configured for launch again at the start of each test. In addition to the formal Mission Test campaigns of the EQM and FM, a number of smaller, informal ~hourslong Mission Tests, have also been carried out by the EIRSAT-1 team during the previous ~months. These tests have been conducted to facilitate final development of the hardware, software, operational procedures and test plans required for the Mission Test. Additionally, these tests have served as a training mechanism for the test operators, who should be familiar with the Mission Control protocols prior to the test.

As can be seen from these details, extensive Mission Testing can be expensive on resources, particularly for a university-class CubeSat project where resources are limited. However, the findings from the survey as well as FYS!'s recommendations give the EIRSAT-1 team confidence that this investment into Mission Testing is extremely worthwhile for mission reliability. Sec. 5 of this works discusses plans to evaluate how this investment into Mission Testing succeeds in practice, given first-hand on-orbit experience with the EIRSAT-1 mission.

4. ADDITIONAL METHODS TO IMPROVE RELIABILITY

This section lists some additional methods and considerations that can be used by CubeSat teams to improve the predicted reliability of their mission:

• Risk/Resource management - Ref. 23 suggest that risk assessments should be carried out at the beginning and regularly throughout a CubeSat project. By doing this, resources can be directed towards mitigating new, existing and evolving risks that can effect the likelihood of mission success.

Examples of risk/resource management carried out by the EIRSAT-1 team to overcome, in particular, risks associated with the CubeSat being the first Irish satellite (i.e. lack of in-house expertise and experience) include:

- Procurement of a Flight Software Development Kit, with flight-ready software components, flight heritage and customer support services,²⁹
- Procurement of COTS components for the standard CubeSat subsystems (see Fig. 1), many of which have flight heritage and are provided by the one supplier,⁹
- Adoption of a model philosophy where a (near-)identical version of the spacecraft (i.e. the EQM) is produced for testing and qualification prior to the final flight-ready version (i.e. the FM) being produced.¹⁹
- **Reliability assessment tools** Different reliability assessment tools exist to provide projects with a way to better quantify the predicted reliability of a project.

Refs. 30,31 list some reliability prediction methodologies and discuss the applicability of these methodologies given the needs and limitations typically associated with CubeSat projects, including university-class projects.³¹ A Reliability Growth Tool developed for CubeSats, which uses test results/failures to make mission reliability estimates (i.e. predicts the number of failures remaining in a system after testing) is also presented by Ref. 30.

• **Project management** - Careful project management is particularly important for university-class Cube-Sat projects, which face unique project challenges such as high turnover of team members as students complete/begin degrees, modules or projects through which they are involved.

To help address these challenges, Ref. 32 conducted interviews with the supervisors of CubeSat teams with multiple successful university-class missions to identify what project management techniques have contributed to their success.

- Lessons learned Many CubeSat teams have disseminated lessons learned on various aspects of their projects with the aim to help other CubeSat teams (e.g. Ref. 33). Additionally, various studies, including the survey-style studies mentioned in this work (e.g. Ref. 23), offer recommendations to CubeSat teams based on their findings. These resources are particularly important for teams developing CubeSats with no in-house experience.
- Test, test! In addition to extensive Mission Testing, improving the quality of other unit- through to system-level testing can be used to improve the predicted reliability of a mission.³⁴

5. FUTURE WORK

This paper presents the initial results of a survey disseminated to CubeSat teams with launch and on-orbit experience. As, to date, ~ 1300 CubeSats have been launched and the number of launches per year is increasing,¹ the potential pool of participants for this study is sizable and steadily growing. Therefore, future work aims to build on the findings presented in this work with input from more survey participants.

In addition to more survey responses, future work also aims to draw on the first-hand experiences of the EIRSAT-1 team. As part of this plan, the EIRSAT-1 team will answer in detail the questions posed to survey participants on their mission operations. The EIRSAT-1 Mission Test plan, which is presented in Sec. 3.2.1, has been developed from the support, recommendations and lessons learned mentioned in this paper. By learning from those with on-orbit experience, the team have aimed to develop test plans that will improve the predicted reliability of the mission and prepare us for launch. In light of first-hand test, launch and on-orbit operations experiences, the EIRSAT-1 team will assess whether this aim was achieved and will highlight what aspects of the Mission Test plan described in Sec. 3.2.1 were most/least beneficial and what should be improved. Using this information, a more comprehensive test guide will be developed to help teams improve the predicted reliability of their CubeSat through Mission Testing.

6. CONCLUSION

This work supports the hypothesis that extensive Mission Testing is an important step towards improving the reliability and performance of a CubeSat prior to launch. In particular, the initial results of a survey, disseminated to CubeSat teams with launch and on-orbit experience, show that Mission Testing is viewed as a valuable tool by teams to improve their likelihood of mission success. However, the results of this survey also show that the scope of Mission Testing being performed varies significantly across CubeSat teams. This finding highlights a need for clearer guidance (e.g. in the form of standards or requirements) for CubeSat teams on the quality of the testing that should be performed to ensure a positive impact on the predicted reliability of their missions. This guidance is particularly important as CubeSat teams typically face constraints on time and resources, and so, allocation of sufficient resources to Mission Testing, with the view that *comprehensive* Mission Testing is an essential part of a CubeSat development project, should be encouraged. By providing details on the EIRSAT-1 test plans, this and future work aims to act as a resource for CubeSat teams to assist the development of, as well as the time required for the development of comprehensive Mission Test plans.

APPENDIX A. MISSION TESTING VS. ON-ORBIT MISSION OPERATIONS SURVEY

This section shows the questions that were presented to survey participants as part of Sections 1) Background, 2) CubeSat Testing and 3) Mission Operations. Participants were not required to answer all questions. However, responses for up to 5 questions, which have been highlighted below with red asterisks, were required to facilitate question logic. Participants where also given the opportunity to expand on answers as desired via *More Information* boxes following each question.

A.1 Background



Figure 6. Section 1) Background survey questions. *Response required to proceed.

A.2 CubeSat Testing



Figure 7. Section 2) CubeSat Testing survey questions. *Response required to proceed. \triangle Note that while some-to-many situations may have been tested by other means (e.g. during Functional or End-to-End Testing), this question specifically relates the situations simulated within a Mission Test.

A.3 Mission Operations



Figure 8. Section 3) Mission Operations survey questions. *Response required to proceed.

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